

Single event tolerance of x-ray silicon-on-insulator pixel sensors

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Abstract. We evaluate the single event tolerance of the x-ray silicon-on-insulator (SOI) pixel sensor named XRPIX, developed for the future x-ray astronomical satellite FORCE. In this work, we measure the cross-section of single event upset (SEU) of the shift register on XRPIX by irradiating heavy ion beams with linear energy transfer (LET) ranging from 0.022 to 68 MeV/(mg/cm²). From the SEU cross-section curve, the saturation cross-section and threshold LET are successfully obtained to be $3.4^{+2.9}_{-0.9} \times 10^{-10}$ cm²/bit and $7.3^{+1.9}_{-3.5}$ MeV/(mg/cm²), respectively. Using these values, the SEU rate in orbit is estimated to be $\lesssim 0.1$ event/year primarily due to the secondary particles induced by cosmic-ray protons. This SEU rate of the shift register on XRPIX is negligible in the FORCE orbit. © The Authors. Published by SPIE under a Creative Commons Attribution 4.0 International License. Distribution or reproduction of this work in whole or in part requires full attribution of the original publication, including its DOI. [DOI: [10.1117/1.JATIS.8.4.046001](https://doi.org/10.1117/1.JATIS.8.4.046001)]

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1 Introduction

The x-ray silicon-on-insulator (SOI) pixel sensor named XRPIX is a monolithic active pixel sensor developed as the main imaging spectrometer onboard the FORCE satellite.¹⁻³ The FORCE satellite is an x-ray astronomical satellite aiming to be launched in the 2030s. It will achieve a broadband x-ray imaging spectroscopy in an energy range of 1 to 79 keV with a high angular resolution better than 15 arcsec. XRPIX is one of the main images of the FORCE satellite and is fabricated with a 0.2 μm fully depleted SOI technology. Utilizing the SOI technology, XRPIX is composed of a high-resistivity Si sensor layer and a low-resistivity CMOS circuit

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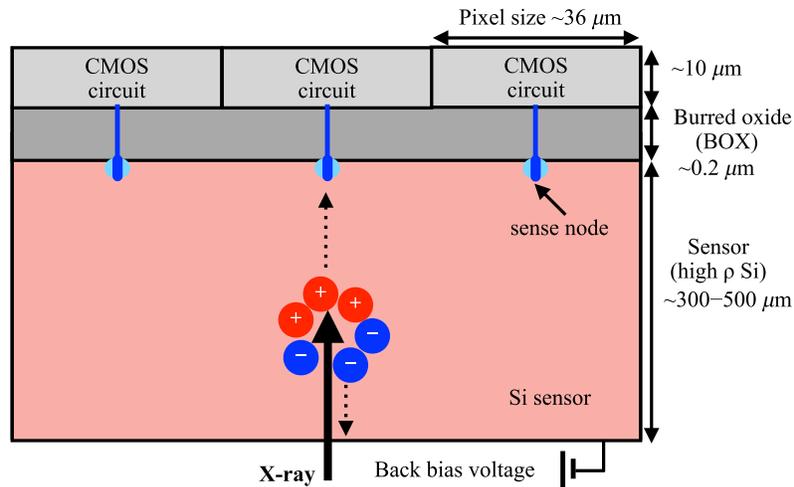


Fig. 1 Schematic view of XRPIX.

layer with a SiO₂ insulator layer in between as shown in Fig. 1. This structure enables to achieve a thick depletion layer as thick as a few hundred micrometers and to implement full CMOS readout circuitry on each pixel. One of the most remarkable characteristics of XRPIX is the event-driven readout, where only signals from the pixels with x-ray events are readout, by using the trigger function implemented in each pixel circuit. It realizes high timing resolution better than $\sim 10 \mu\text{s}$, which enables an extremely low background observation by adopting the anticoincidence technique.

The tolerance to single event effect (SEE) should be considered in the development of the CMOS integrated circuits for space use. The SEE is a radiation effect caused by a strike of a single energetic particle. There are two major categories in the SEE: single event upset (SEU) and single event latch-up (SEL).⁴ In the SEU, the logical state of a digital circuit is changed by the free charge generated by the incident particle. It is not destructive but causes the malfunction of the circuit. On the other hand, the SEL is potentially destructive because it results in a large current by turning on a parasitic thyristor structure in the CMOS circuit.

Since the CMOS circuit of XRPIX is SOI-CMOS, single event tolerance should be better than that of bulk CMOS devices.⁵ In principle, the SEL cannot occur because there is no parasitic thyristor in the SOI-CMOS. The SEU is also mitigated because the SEU-sensitive volume is reduced in the SOI-CMOS. However, it is unknown whether the SEU tolerance of XRPIX is enough for the FORCE satellite. In particular, in the FORCE satellite, the CMOS circuit of the flight model of XRPIX will contain a lot of shift registers storing operational parameters. If SEU will frequently occur in such shift registers, it would have a strong impact on the operations of XRPIX in orbit. Thus, in this work, we measured the SEU cross-section curve of the shift register on XRPIX for the first time, and quantitatively evaluated its SEU tolerance. The rest of the paper is organized as follows. In Sec. 2, we describe the details of the heavy-ion irradiation experiment. In Sec. 3, we present the main results, and estimate the SEU rate in orbit in Sec. 4. We conclude in Sec. 5.

2 Heavy-ion Irradiation Experiment

2.1 Test Device: XRPIX8

We irradiated heavy ions to XRPIX at the Heavy Ion Medical Accelerator in Chiba (HIMAC) in the National Institute of Radiological Sciences. The test device used in this experiment was the current prototype of the XRPIX series named “XRPIX8.” XRPIX8 has a *p*-type sensor layer with a thickness of 300 μm. It has 96 × 96 pixels with a pixel size of 36 × 36 μm², so that the sensitive area is $\sim 3.5 \times 3.5 \text{ mm}^2$. Adopting the pinned depleted diode (PDD) structure, XRPIX8

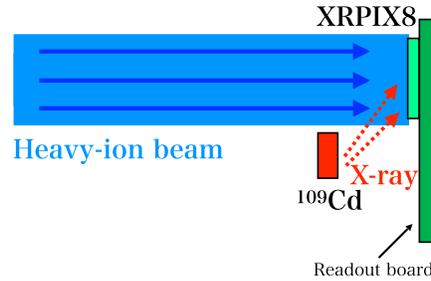


Fig. 2 Schematic picture of the experimental setup of heavy-ion irradiation of XRPIX.

has an energy resolution as good as XRPIX6E, which achieved the best spectral performance among all the XRPIX series.⁶

In this experiment, we evaluated the single event tolerance of shift registers equipped in the on-chip peripheral circuits in XRPIX8.⁷ The shift register is composed of a D-type flip-flop circuit, and used as a trigger mask for ignoring triggers from noisy rows or columns. By writing a mask pattern to the shift register, trigger signals from the corresponding row or column are ignored. Since XRPIX8 has 96×96 pixels, there are two 96-bit shift registers for the trigger masks of rows and columns in XRPIX8.

To monitor the SEU in the shift registers, we periodically read out the values in the shift registers on XRPIX during the heavy-ion irradiation. For each readout, SEU was judged by comparing the values in the shift register with the written values. In addition to the SEU monitoring, we monitored the current consumption in XRPIX by applying voltages with source meters to detect SEL if it occurs.

During the heavy-ion irradiation, XRPIX was also irradiated with 22-keV x-rays from a radioisotope ^{109}Cd as shown in Fig. 2. While the experiments with ions above 100 MeV/u were performed in the air, the experimental setup was located in a vacuum chamber for 6-MeV/u beams to avoid energy loss of the beam in the air. The XRPIX sensor was operated under a back-bias voltage of -25 V, which creates approximately $100 \mu\text{m}$ of the depletion layer. Thus, both the heavy ions and x-rays were measured with XRPIX during the experiment.

2.2 Heavy Ion Beam

To evaluate the SEU tolerance, it is necessary to determine the SEU cross-section as a function of linear energy transfer (LET). Therefore, we adopted a variety of heavy ion beams with the LET values ranging from 0.022 to 68 MeV/(mg/cm²) as listed in Table 1. The LET values were estimated by using Geant4 simulation.⁸⁻¹⁰ The total fluence at each LET was $6 \times 10^7 - 9 \times 10^9 \text{ cm}^{-2}$, and the flux was $9 \times 10^3 - 2 \times 10^6 \text{ cm}^{-2} \text{ s}^{-1}$. Since the XRPIX device with

Table 1 Summary of the heavy-ion irradiation.

Ion	Energy (MeV/u)	LET [MeV/(mg/cm ²)]	Fluence (particles/cm ²)
He	100	0.022	2.96×10^9
H	6	0.051	8.65×10^9
Si	400	0.42	3.52×10^9
Kr	200	4.2	1.88×10^8
Xe	200	9.3	1.85×10^8
Fe	6	25	2.34×10^8
Xe	6	68	6.11×10^7

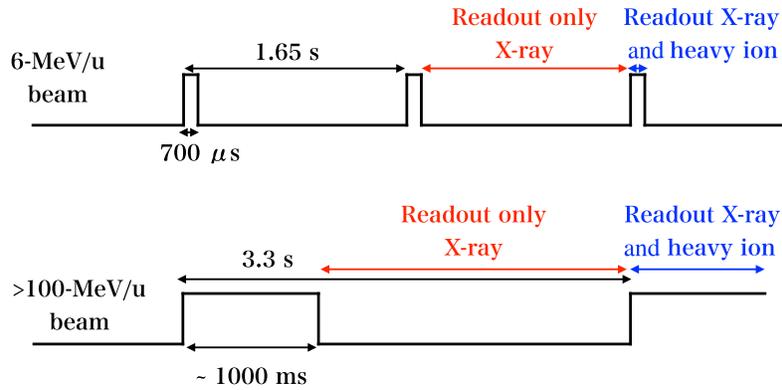


Fig. 3 Typical time profiles of the heavy-ion beams.

PDD structure was established to work up to ~ 100 krad $\simeq 6.3 \times 10^9$ MeV/mg in our previous work,¹¹ we regulated the total fluence at each LET not to exceed it.

Figure 3 shows typical time profiles of the heavy-ion beams. 700- μ s pulses were irradiated with a period of 1.65 s, while 1000-ms pulses (depending on ion) with a 3.3-s period. Since x-ray is also irradiated as described in Sec. 2.1, only x-ray was readout during the beam-off phase, while both x-ray and heavy-ions were readout during the beam-on phase.

3 Results of the Irradiation Experiment

3.1 Performance of XRPIX During the Heavy-Ion Irradiation

For the demonstration purpose, we show the measured spectra with XRPIX during the heavy-ion irradiation in Fig. 4. As shown in the figure, even in the beam-on phase, we were able to measure x-rays from ^{109}Cd above the continuum by the heavy ions. Also, in the beam-off phase, the spectral performance is almost the same as that before the irradiation. Therefore, as demonstrated in Fig. 4, XRPIX works in the event-driven readout mode without any major malfunctions even with the heavy-ion irradiation.

According to the current consumption monitoring of XRPIX during the irradiation, we did not find any signs of the SEL. Throughout the experiment, the current consumption was kept at steady values. It is a very plausible result because SEL never occurs in the SOI-CMOS devices in principle as described in Sec. 1.

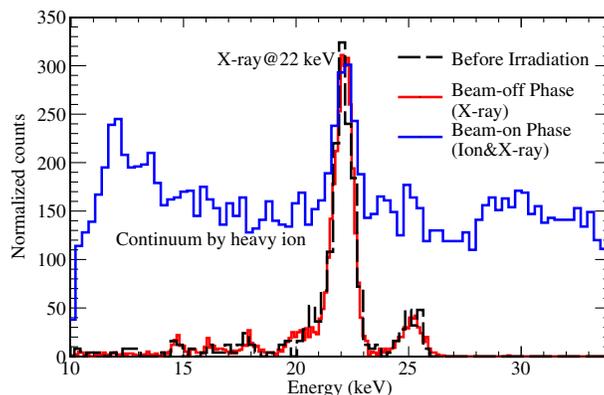


Fig. 4 ^{109}Cd spectra measured with XRPIX during the heavy-ion irradiation (red/blue) compared with that before the irradiation (black).

3.2 SEU Cross-Section Curve

Figure 5 shows the SEU cross-section curve $\sigma(L)$ as a function of the LET value L obtained by the heavy-ion irradiation experiment. In general, the SEU cross-section curve is described as a Weibull function¹²

$$\sigma(L) = \begin{cases} \sigma_{\infty} \left[1 - \exp\left(-\frac{L-L_{\text{th}}}{W}\right) \right] & (L \geq L_{\text{th}}), \\ 0 & (L < L_{\text{th}}), \end{cases} \quad (1)$$

where σ_{∞} is the saturation cross-section, L_{th} is the threshold LET, and W is the curve width. Thus, by fitting with the Weibull function, we can obtain these characteristic parameters of the SEU cross-section curve.

Although we successfully detected the SEUs above $\sim 9.5 \text{ MeV}/(\text{mg}/\text{cm}^2)$, no SEU was detected below $\sim 4.5 \text{ MeV}/(\text{mg}/\text{cm}^2)$, where only upper limits are shown as a triangle in the figure. Thus, to utilize the data with no SEU detection, we adopted a maximum likelihood method of the binomial distribution for the fitting. Since the SEU event is a Bernoulli trial where the stored bit is flipped or not, the number of SEU k follows a binomial distribution

$$P(x) = \binom{n}{k} p^k (1-p)^{n-k}, \quad (2)$$

where p is the probability of SEU and n is the number of incidence particles. Although it can be approximated to the Poisson distribution in our experiment, we implemented the likelihood function of the binomial distribution for wide applicability. Thus, we minimized the likelihood function defined as

$$-2 \ln L = -2 \left[\ln \binom{n}{k} + k \ln p + (n-k) \ln(1-p) \right], \quad (3)$$

for the SEU probability p expected in the model, and obtained the best-fit parameters of the Weibull function. We also estimated the uncertainties of parameters as the intervals where the difference of $-2 \ln L$ from the best-fit was below unity. As the result of fitting, we obtained the threshold LET of $L_{\text{th}} = 7.3^{+1.9}_{-3.5} \text{ MeV}/(\text{mg}/\text{cm}^2)$, the saturation cross-section of $\sigma_{\infty} = 3.4^{+2.9}_{-0.9} \times 10^{-10} \text{ cm}^2/\text{bit}$, and the curve width of $W < 35 \text{ MeV}/(\text{mg}/\text{cm}^2)$ (68% upper limit).

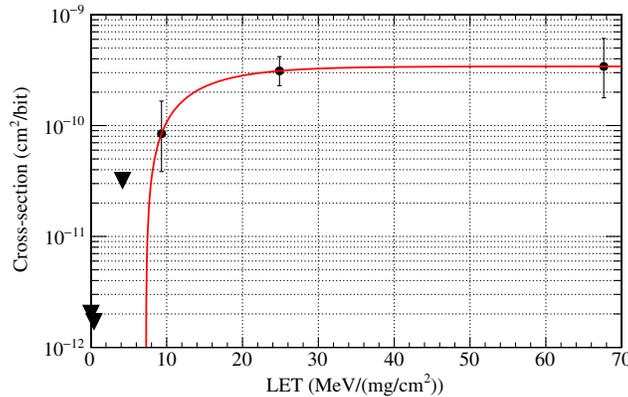


Fig. 5 SEU cross-section curve of the shift register on XRPIX (black points) and the best-fit Weibull function (red solid curve). Filled circles indicate the experimental data points with SEU detection, while triangles indicate 68% upper limits without SEU detection.

4 Estimation of SEU Rate in Orbit

Utilizing the obtained saturation cross-section and threshold LET, we estimated the SEU rate of the shift registers in XRPIX in the FORCE orbit. We assumed radiation conditions in the solar maximum in this section. Since the flight model of XRPIX will have $\sim 1000 \times 1000$ pixels, the total number of bits in the shift registers for the trigger mask will increase to ~ 2000 . In addition, to reduce the size of the detector system, the flight model is planned to be equipped with digital circuitry, which is currently in the part of the readout board. Although the number of bits implemented in the on-chip digital circuitry is not yet decided, it would be $< 10^4$ bits. Thus, we used this conservative value as the total number of bits in the shift register to estimate the SEU rate.

We first estimated the SEU rate induced by heavy ions in orbit. According to the Space Environment Information System (SPENVIS),¹³ the heavy-ion flux above the threshold LET [$L_{\text{th}} = 7.3_{-3.5}^{+1.9}$ MeV/(mg/cm²)] is $\lesssim 2 \times 10^{-9}$ particles/cm²/s in the FORCE orbit. As the FORCE orbit, we assumed the low earth orbit with an altitude of 550 km and an inclination of 30 deg.³ By multiplying this flux, the assumed number of bits, and the measured cross-section, we estimated the SEU rate due to the heavy ions to be as small as $\sim 10^{-7}$ event/year.

In addition to the SEU by heavy ions, we also considered the SEU effect due to secondary particles generated by the incident cosmic-ray protons. This effect has a significant contribution to the SEU in the FORCE orbit. Although the cosmic-ray protons deposit small energies compared with the heavy ions, they can cause nuclear reactions with the device material. This nuclear reaction generates the secondary particles, and they have large energy deposits on the device. Thus, we need to estimate the SEU due to the secondary particles generated by the cosmic-ray protons.

According to the simple formula proposed in Barak et al.,¹⁴ the SEU cross-section due to the proton-induced secondary particles is estimated to be $\sim 10^{-15}$ cm²/bit at maximum. Since the flux of geomagnetically trapped protons is 1×10^2 particle/cm²/s in the FORCE orbit, the proton-induced SEU rate is calculated to be $\lesssim 0.1$ event/year. The actual SEU rate should be much smaller than this estimation in the FORCE satellite because XRPIX is planned to be surrounded by ~ 30 -mm thick BGO shields. Therefore, the SEU rate of the shift registers on XRPIX is negligible in the FORCE satellite.

The SEU does not have a significant impact on the operation of XRPIX even if it occurs in orbit, though it was found to be a very rare event in this work. Since the SEU is not a destructive event, if it occurs in space, it can be fixed by rewriting parameters to registers. It could cause problems in celestial observation and satellite operation if the SEU frequently occurs. However, in the case of low SEU probability as in this work, the effect of the SEU is avoidable by regularly rewriting the registers. Therefore, we can conclude that the XRPIX has a sufficient SEU tolerance for the use onboard the FORCE satellite.

5 Conclusions

We evaluated the SEU tolerance of XRPIX for the first time by irradiating the heavy-ion beams. In this experiment, we found that XRPIX did not have any major malfunctions during the irradiation. Also, we successfully estimated the threshold LET and saturation cross-section from the experimental results. According to these values, the SEU rate of XRPIX in orbit is as rare as $\lesssim 0.1$ event/year. Therefore, we found that XRPIX has a sufficient SEU tolerance for the use onboard the FORCE satellite.

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